



Precision in Motion.

Geo-iNAV[®] Tactical GPS/INS Navigation System

The Smallest & Lightest High-Performance GPS/INS Available



Geo-iNAV[®] **Tactical** is a fully-integrated GPS-aided inertial navigation system that utilizes a high-stability Quartz MEMS IMU to provide a high-performance navigation solution. Geo-iNAV[®] Tactical is offered in several configurations designed to meet a wide range of requirements, and is available for both commercial and military applications:

Geo-iNAV[®] Tactical Commercial: Designed for civilian navigation applications (no ITAR restrictions) **Geo-iNAV[®] Tactical SAASM:** Designed for applications that have a military SAASM GPS requirement

Geo-iNAV[®] Tactical Features

- Centimeter-level position accuracy (dual-frequency RTK configuration)
- GPS Processing with Precise Instantaneous Network positioning based on Geodetics' Epoch-by-Epoch[®] technology
- Tight-coupling with Geodetics' Extended Kalman Filter
- Full post-processing support with Geo-PostProcessing tools
- Support for low, medium and high-dynamic platforms
- In-motion dynamic alignment
- Mil-spec ruggedization
- Optional TDMA wireless data-link

Geo-iNAV[®] Tactical Applications

- UAV & UGV navigation and control
- Military & defense security
- Robotic & ROV control
- Oil & gas exploration
- Mobile mapping systems & photogrammetry
- Transportation safety & maintenance
- Construction & structural management



The Geo-iNAV[®] Tactical INS is a joint venture between Geodetics, Inc. and Epson Electronics America. For all sales and technical-support related questions, please contact Geodetics, Inc. directly via:

P: 858.729.0872 F: 858.729.0874 E: info@geodetics.com

Geo-iNAV[®] Tactical Product Configuration Options



Geo-iNAV[®] Tactical GPS Options

GPS Options	Processing Options	Typical Position, Velocity, Acceleration, Attitude Accuracy (RMS) ¹		
		Horizontal , Vertical Pos.	Velocity , Acceleration	Roll, Pitch / Heading
Commercial	LI Standalone	l.5 m , 2.5 m	0.1 m/s , 0.15 m/s ²	0.2° / 0.5°
	LI/L2 RTK	5 cm , 10 cm	0.02 m/s , 0.1 m/s ²	0.1° / 0.3°
SAASM	L1/L2 Standalone	I.0 m , 2.0 m	0.1 m/s , 0.15 m/s ²	0.2° / 0.5°
	LI/L2 RTK	5 cm ,10 cm	0.02 m/s , 0.1 m/s ²	0.1° / 0.3°

Geo-iNAV[®] Tactical IMU Technical Specifications

Parameter	Epson M-G362	Epson M-G352
Gyroscope Dynamic Range	±150°/sec.	±440°/sec.
Gyroscope Bias In-Run Stability (1σ)	3°/hr.	6°/hr.
Gyroscope Angle Random Walk (1σ)	0.l°/√hr.	0.2°/√hr.
Accelerometer Dynamic Range	±3g	±6g
Accelerometer Bias In-Run Stability (1σ)	<0.1mg	<0.1mg
Accelerometer Velocity Random Walk (1σ)	0.04(m/sec)/√hr.	0.04(m/sec)/√hr.

Maximum Drift of the Navigation Solution (Position, Velocity, Attitude) after GPS Outages

Outage	Position (Horizontal , Vertical)	Velocity (Horizontal , Vertical)	Attitude (Roll, Pitch / Heading)
30 seconds	2.0 m , 1.5 m	0.2 m/s , 0.1 m/s	0.02° / 0.05 °
60 seconds	8.0 m , 5.5 m	0.4 m/s , 0.5 m/s	0.05° / 0.1°

Geo-iNAV[®] Tactical Technical Specifications

Parameter	Commercial Configurations	SAASM Configurations	
Size / Weight / Power	33.8 in ³ (4.73x3.95x1.81) / 20 oz. / 10 – 30 VDC @ 2 Amps min.		
Environmental	MIL-810E, MIL-461 Compliant		
Temperature Range	Specified: -20°C to +60°C Operating: -40°C to +70°C		
Interfaces	External power connector, TNC GPS antenna connector, 1 Ethernet data port, 3 RS-232 serial ports, 1PPS output, 4 status LEDs.	External power connector, TNC GPS antenna connector, I Ethernet data port, 3 RS-232 serial ports, IPPS output, 4 status LEDs, SAASM Keyload Connector, SAASM Zeroize button.	
Real-Time Data Output	Navigation solutions at 125 Hz. available via Ethernet, RS-232 or optional wireless data-link.		
Data Recording/Logging	Navigation solutions, raw GPS & IMU data (for post-processing with Geo-PostProcessing tools).		
Wireless Communications	Optional TDMA data-link (various frequencies available), Point-to-Point mode available.		
GPS Frequency Tracking	LI & LI/L2	LI & L2 (P/Y Code)	
Key Loading	N/A	DS101	
Safety & Diagnostics	Internal safety and monitoring systems. Internal BIT with operator notification.		

¹Accuracy is dependent on GPS satellite system performance, ionospheric conditions, satellite visibility, data-link and other factors.

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